



Issued: 6 February 2025

Robinson Helicopters Time-Controlled Components and Service Life Limited Components

This Safety Notice contains recommendations regarding operational safety.

Recipients must ensure that this Notice is copied to all members of their staff who need to take appropriate action or who may have an interest in the information (including any 'in-house' or contracted maintenance organisations and relevant outside contractors).

Applicability:				
Aerodromes:	Not primarily affected			
Air Traffic:	Not primarily affected			
Airspace:	: Not primarily affected			
Airworthiness:	iness: Part-CAMO/Part-M, Part-CAO, Part-ML, aircraft owners, licenced aircraft engineers			
Flight Operations:	erations: Not primarily affected			
Licensed/Unlicensed Personnel:	Not primarily affected			
Affected Products/TCDS's	UK adopted EASA.IM.R.120 Issue 5 UK adopted EASA.IM.R.121 Issue 6 UK.RC.R.00078 Issue 1			

1 Introduction

1.1 The UK CAA have become aware that the Robinson Helicopter's Instructions for Continued Airworthiness (ICA) for Robinson R22, R44 and R66 Helicopters may be misinterpreted, particularly in respect of the correct control and tracking of time-controlled components, as required by Part-M in M.A.503(a), also referred to as service life limited components in Part-ML in ML.A.503(a)(2). These commonly occur following updates to the ICA requiring retrospective actions resulting from the amended life limit or service life limit.

Failing to adhere to the requirements of the ICA could potentially lead to an unsafe flight condition due to installed components being installed for a longer period than originally intended in accordance with the approved maintenance data.

2 Action to be Taken

2.1 Owners, operators, licenced engineers and contracted Part-CAMO and Part-CAO organisations are reminded of their responsibilities, as required by M.A.302(d)(b), M.A.305(d)(2), ML.A.302(c)(2) and ML.A.305(d)(4), to control these components.

With immediate effect from the publishing date of this Safety Notice, owners, operators, licenced engineers and contracted Part-CAMO and Part-CAO organisations are requested to review their maintenance programmes to ensure components are being managed correctly.

Where an incorrect management of any components is identified, all necessary steps should be taken to achieve compliance with the Robinson Helicopter ICA. Should a helicopter be flying with components identified as beyond their recommended limitation, the subject helicopter should be immediately grounded, a Mandatory Occurrence Report raised together with a notification to the CAA, in order to co-ordinate a recovery plan. If any repositioning is required to return a helicopter to a place where maintenance can be carried out, then a Permit to Fly request should be made to the CAA. A supporting letter from the Type Certificate Holder, Robinson Helicopters, detailing the components beyond their limitation should be included.

3 Queries

3.1 Any queries or requests for further guidance as a result of this communication should be addressed to airworthiness@caa.co.uk or where applicable, an organisation assigned CAA Airworthiness Surveyor.

4 Cancellation

4.1 This Safety Notice will remain in force until withdrawn.

5 Definitions

5.1 GM M.A.305

"(e) The term 'time-controlled components' embraces any component for which the maintenance schedule of the aircraft maintenance programme requires periodically the removal for maintenance to be performed in an appropriate approved organisation for maintenance in components (workshop) to return the component to a specified standard, the replacement of sub-components of the assembly by new ones, or the inspection or test of component's performance, after a service period controlled at component level in accordance with the specified airworthiness limitation defined in accordance with UK Regulation (EU) No. 784/2012, in any of the applicable parameters."

5.2 ML.A.503

- "(a) The term 'service life-limited components' contains the following components:
- (1) components subject to a certified life limit after which the components should be retired, and;

(2) components subject to a service life limit after which the components shall undergo maintenance to restore their serviceability."

Appendix 1 Clarification on Requirements for Life Tracking

The type certificate data sheet (TCDS) sets out the requirement to satisfy the ICA, as required by the Certification Basis for the product. In this example, for the Robinson R44, the UK adopted EASA TCDS, IM.R.121:

22. Life-limited Parts	See Robinson Maintenance Manual and Instructions for Continued Airworthiness (RTR 460).			
	Retirement times are listed in the approved			
	"Airworthiness Limitations" section of Chapter 3.			
•	Safety Agency, 2017. All rights reserved. ISO9001 certified. Page 1 trolled. Confirm revision status through the EASA-Internet/Intranet.			
TCDS No.: EASA.IM.R.121	R44			
TCDS No.: EASA.IM.R.121 Issue: 06	R44 Date: 28 February 2017			
Issue: 06				
Issue: 06	Date: 28 February 2017 Robinson Helicopter Company R44 II Rotorcraft Flight Manual, RTR 462, dated 3 October 2002, with revisions			
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Several sections of the Maintenance Manual detail the requirement for tracking components. Section 1.102 of the Robinson R44 Maintenance Manual details components to be inspected and/or replaced at either 12 years or 2200 hrs, whichever occurs first:

2 Additional	Component Maintenance		1.102 Additional Co	mponent Maintenance (continued)	
	NOT	F	B. 2200/2400 Hou	rs	
additi	nanufactured parts not lis	ted in § 1.102 as requiring e, or replacement per § 3.300,	2200 hours (R	indicated on the following component 14 S/Ns 0001 thru 9999 & R44 II S/I let S/Ns 30001 & subsequent) time in	Ns 10001 thru 29999) or <u>240</u>
12 YEARS			Part Number	Description	Action
Perform acti	on indicated on the followin	g components when they have accumulated			
12 years cal	endar time and less than 220	0 hours (R44 S/Ns 0001 thru 9999 & R44 II R44 Cadet S/Ns 30001 & subsequent) time in	A120-3 A130-21	Tail Rotor Bellcrank Spacer (at R44 and R44 Cadet powerplant controls)	Replace with new. Replace with new.
service since	new, since last overhaul, or	since last 12-year maintenance:	A130-48	Spacer	Replace with new, per R44 SL-80
			A190-3	V-Belt Set	Replace with new.
Part Number	Description	Action	A336-6, -9	Push-Pull Tube - Throttle (R44 II)	Beplace with new.
A120-3	Tail Rotor Bellcrank Assembly	Replace with new.	A462-4	Fitting	Replace with new.
A130-48	Spacer	Replace with new, per R44 SL-80.	A522-7	Control Cable – Mixture (carburetor)	Replace with new.
A190-3	V-Belt Set	Replace with new.	A522-13	Control Cable – Mixture (fuel control)	Replace with new.
A336-6 or -9	Push-Pull Tube, Throttle (R44	Visually inspect. If exterior corrosion is evident,	A595-1	Seal - Vertical Firewall (neoprene)	Beplace with new.
	ID .	record length, disassemble, and inspect tube	A595-2	Seal - Vertical Firewall (Teflon®)	Replace with new.
	attest of the second second	interior. Repair or replace as required.	A650-2 or -4	Fitting - MRGB Mount	Visually inspect, including bore.
A462-4 A650-2 or -4	Fitting (mixture control arm) Fitting (MRGB mount)	Visually inspect. Replace if worn or corroded. Visually inspect, including bore. Replace if worn or corroded.	1000 2 01 4	Tung meenean	Replace if worn or corroded. Magnetic particle inspect per
A785-6	Hose (bulkhead to MRGB)	Beplace with new.	1 200 00	Tube (aux fuel pump drain)	§ 23-41. Replace with new.
A785-7	Hose (alternator cooling)	Replace with new.	A729-33	Hose (bulkhead to MRGB)	Replace with new.
A785-10	Hose (carb heat scoop to	Beplace with new.	A785-6 A785-7	Hose (alternator cooling)	Replace with new.
	airbox)		A785-10	Hose (are heat scoop to airbox)	Replace with new.
A785-11	Hose (engine LH cowling to	Replace with new.	A785-10 A785-11	Hose (engine LH cowling to airbox)	Replace with new.
A785-12	airbox) Hose (acroll to muffler shroud)	Beplace with new	A785-12	Hose (scroll to muffler shroud)	Replace with new.
A785-12 A785-13	Hose (scroll to muttler shroud) Hose (muffler shroud to cabin	Replace with new.	A785-12 A785-13	Hose (scroll to marrier stroud) Hose (muffler shroud to cabin heat inlet)	Replace with new.
A765-13	heat inlet)	Replace with new.	A785-16	Hose (scroll to MRGB)	Replace with new.
A785-16	Hose (scroll to MRGB)	Replace with new.	A785-10 A785-17	Hose (scroll to bulkhead)	Replace with new.
A785-17	Hose (scroll to bulkhead)	Replace with new.	A785-19	Hose (magneto cooling)	Replace with new.
A785-19	Hose (magneto cooling)	Replace with new.	A785-28	Hose (bulkhead to hydraulic reservoir)	Replace with new.
A785-28	Hose (bulkhead to hydraulic reservoir)	Replace with new.	A785-31	Hose (R44 II engine air intake) Elastic Cord – Tail Rotor	Replace with new. Replace with new. Replace with new. Dash number
A785-31	Hose (R44 II engine air intake)	Replace with new.	A918-1 thru -8	Elastic Cord – Tall Hotor	selected during flight test evaluation
A785-32	Hose (alternator cooling)	Replace with new.	A947-2	Flex Plate Assembly (intermediate)	Replace with new.
A918-1 thru	Elastic Cord - Tail Rotor	Replace with new. Dash number is selected during	B173-2, -3, or -6		Replace with new.
-8		flight test evaluation.	B173-4	V-Belt - A/C Compressor Drive	Replace with new.
A947-2	Flex Plate Assembly (bonded)	Visually inspect with 10x magnification. Replace	B277-024	Clamp	Replace with new.
		if any bonded washer evidences separation (8 places). Replace if corrosion is evident.	B277-036	Clamp	Replace with new.
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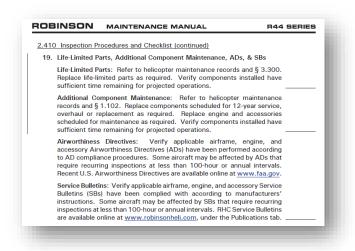
NOTE

If a component or an inspection is scheduled for hourly and calendar intervals, comply with whichever requirement comes first, then reset interval unless otherwise specified.

The table in section 1.102 for the 12 year or 2200-hour inspection includes additional components that are not listed within the Chapter 3 of the Maintenance Manual:

300 Airworthiness Limi	tations	
The Airworthiness Limita	tions Section is FAA approved and specifi der 14 CFR §§ 43.16 and 91.403, unles	
R44 and R44 II helicopt	tigue life-limited parts. The first list (this ers. The second list (following page) pr d for the two-seat R44 Cadet configurat	ovides increased service
R44 and R44 II Fatigue	e Life-Limited Parts	
Part Number	Description	Maximum Service Life
 C023-1		2000 Hours
C016-2 ⁴ , -5 ⁴ , & -7 C020-1 & -2 C029-1, -2, & -3 C030-1 C044-1 C044-1 C146-1 C146-2 C146-1 C158-1 C158-1 C263-1 & -2 C264-1 & -2 C545-1 C545-2 C545-2	Main Rotor Blade Upper Frame Tail Rotor Blade Tail Rotor Hub Horizontal Stabilzer, Rev L & Prior Gear Set, Main Gearbox Pinion, Main Gearbox Main Rotor Spindle Tail Rotor Drive Shaft Sump, Main Gearbox Housing, Main Gearbox Gear Set, Tail Gearbox Bearing Set, CO17-6 Swashplate Tail Rotor Guard	2200 Hours or 12 years ¹ 2200 Hours 2200 Hours or 12 years ¹ 2200 Hours ² 2200 Hours ² 2200 Hours 2200 Hours ² 2200 Hours ² 2200 Hours 2200 Hours 2200 Hours 2200 Hours 2200 Hours ² 2200 Hours ² 2200 Hours ² 2200 Hours ² 2200 Hours ²
C023-2, -3, -4, -14, &-15 C044-1 C198-1 &-2 C251-1 C319-3 ⁴ C320-1 ⁴ C337-1 ⁴ D196-1 P505-2 ¹ Whichever limit occurs firr ² Maximum service life ia 2 ³ Maximum service life ia 4 ⁴ Obsolete due to FAA AD	Tailcone Assembly, Rev N & Subsequent Tailcone Assembly Horizontal Stabilizer, Rev M & Subsequent Lower Swashplate Main Rotor Shaft Cyclic Torque Tube Cyclic Stick Jackhahft Tail Rotor Drive Shaft Horizontal Stabilizer t. Calendar time starts on date of original RHC-is 000 hours if part is, or ever has been, installed or 000 hours if part is, or ever has been, installed or 2001 August 19.	4400 Hours 4400 Hours 4400 Hours ² 4400 Hours 4400 Hours 4400 Hours 4400 Hours 4400 Hours 4400 Hours 4400 Hours ² 3804 Arworthiness Approval. an R66 helicopter.
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Specifically, within the 100-hr (Annual) Scheduled Inspection, there is a requirement for the verification of component status, including verification that the installed components have sufficient life remaining in line with Table 1.102:



It should be noted that Table 1.102 is a stand-alone requirement for the Additional Component Maintenance independent of any other scheduled check.