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# TYPE-CERTIFICATE DATA SHEET

No. P.020

**For Propeller**  
AV-804 series propellers

**Type Certificate Holder**  
Avia Propeller Ltd.  
Beranových 65/666  
199 00 Praha 9  
Czech Republic

For Models:  
AV-804-1



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## **I. General**

### **1. Type/ Models**

AV-804 / AV-804-1

### **2. Type Certificate Holder**

Avia Propeller Ltd.  
Beranových 65/666  
199 00 Praha 9  
Czech Republic

Design Organisation Approval No.: EASA.21J.072

### **3. Manufacturer**

Avia Propeller Ltd.  
Beranových 65/666  
199 00 Praha 9  
Czech Republic

### **4. Date of Application**

AV-804-1
20 May 2015

### **5. EASA Type Certification Date**

AV-804-1
08 February 2018

## **II. Certification Basis**

### **1. Reference Date for determining the applicable airworthiness requirements**

20 May 2015

### **2. EASA Certification Basis**

#### **2.1. Airworthiness Standards**

CS-P Amendment 1 as published with ED Decision 2006/09/R dated 16 November 2006 except the CS-P 550 and CS-P 560 as allowed by CS-P 10(b), see note 1.

#### **2.2. Special Conditions**

None

#### **2.3. Equivalent Safety Findings**

None

#### **2.4. Deviations**

None



### **III. Technical Characteristics**

#### **1. Type Design Definition**

Each design configuration is defined by a main assembly drawing and an appropriate parts list.

The AV-804-1-( )-C-F-R(W) propeller model covers the following design configurations.

Design Configuration "Constant Speed, Feather, Reverse (System Walter )" "

Drawing No. 117-0000 dated May 5, 2015 (\*1)

Parts List No. R-117-0000 dated May 5, 2015 (\*1)

(\*1)effective is the declared issue or a later approved revision.

#### **2. Description**

The AV-804-1 propeller model is 4-blade variable pitch propeller with a hydraulically operated blade pitch change mechanism providing the operation "Constant speed", "Feather", and "Reverse".

The hub is milled out of aluminium alloy.

The blades are forged and milled out of aluminium alloy.

Optionally the propeller may have installed a spinner and ice protection equipment.

#### **3. Equipment**

Spinner: according to Avia Propeller Service Bulletin No. 2M and all later approved versions.

Governor: according to Avia Propeller Service Bulletin No. 3N and all later approved versions.

Ice Protection: according to Avia Propeller Service Bulletin No. 4J and all later approved versions.

#### **4. Dimensions**

Propeller diameter: max. 2500 mm (98,4 in)

#### **5. Weight**

AV-804-1 Propeller Design Configuration

"Constant speed, Feather, Reverse",

propeller spinner incl.: approx.. 80,0 kg (176,4 lb)

#### **6. Hub/ Blade- Combinations**

Hub	Blade - Type
AV-804-1	-441

#### **7. Control System**

Propeller governors as listed in Avia Propeller Service Bulletin No. 3N and all later approved versions.

All governors and propeller control systems must be approved as part of the aircraft installation regardless of manufacturer.

#### **8. Adaptation to Engine**

ARP 880 Flange.

#### **9. Direction of Rotation**

Right-hand tractor (viewed in flight direction)



#### **IV. Operating Limitations**

##### **1. Approved Installations**

Specific installation not yet defined. The general suitability of all AV-804 propeller models for a given aircraft/engine combination must be demonstrated within the scope of the type certification of the aircraft.

Refer to Avia Propeller Service Bulletin No.5G and all later approved versions.

##### **2. Maximum Take Off Power and Speed**

635 kW (850 HP)  
1950 RPM

##### **3. Maximum Continuous Power and Speed**

635 kW (850 HP)  
1950 RPM

##### **4. Propeller Pitch Angle**

Maximum pitch change range 105° - measured at 75% radius station

#### **V. Operating and Service Instructions**

Operation and Installation Manual	P/N EN-1320 Date of Latest Issue/Revision Issue 4, Rev. July 27, 2018 (*)
Overhaul Manual	P/N EN-1291 Date of Latest Issue/Revision Issue 4, Rev. June 25, 2018 (*)
Overhaul Manual for Metal Blades	P/N EN-1370 Date of Latest Issue/Revision Issue 5, Rev. March 12, 2018 (*)
Service Bulletins	as noted in the current List of Service Bulletins

(\*) effective is the declared issue or a later approved revision



## VI. Notes

1. This Propeller has been certificated in accordance with CS-P subparts A, B and C. Compliance with the requirements of Subpart D, which is specific to each aircraft installation, has not yet been demonstrated.
2. The EASA approved Airworthiness Limitations Section (ALS) of the Instructions for Continued Airworthiness is published in the applicable "Propeller Operation and Installation Manual" document, chapter 1 "Airworthiness Limitations".
3. The suitability of a propeller for certain aircraft/engine combination must be demonstrated within the scope of the type certification of the aircraft.
4. Propeller designation system:

Hub /Blade  
AV - 804 - 1 - E - ( ) - ( ) - ( ) - ( ) / ( ) ( ) 250 - 441 ( )  
1 2 3 4 5 6 7 8 9 10 11 1 2 3 4 5

### Hub

- 1 Avia Propeller (manufacturer)
- 2 V - Variable Pitch Propeller
- 3 Blade Root Type
- 4 Number of Blade
- 5 No. of variant of the propeller model
- 6 code letter for flange type
  - B = AS-127-D, SAE No.2 mod., ½ inch - 20 UNF bolts
  - D = ARP 502
  - E = ARP 880
  - K = M14 Flange
- 7 code letter for counterweights
  - blank = no or small counterweights for pitch change forces to decrease pitch
  - C = counterweights for pitch change forces to increase pitch
- 8 code letter for feather provision
  - blank = no feather position possible
  - F = feather position installed
- 9 code letter for reverse provision
  - blank = no reverse position possible
  - R = reverse position installed



- 10 code letter for reverse system  
(W) = System Walter
- 11 code letter for design changes  
small letter for changes which do not affect interchangeability  
capital letter for changes which restrict or exclude interchangeability

### Blade

- 1 code letter for position of pitch change pin
  - blank = pitch change pin position for pitch change forces to decrease pitch
  - C = pitch change pin position for pitch change forces to increase pitch
  - CF = pitch change pin position for feather provision; pitch change forces to increase pitch
  - CR = pitch change pin position for reverse provision; pitch change forces to increase pitch
  - CFR = pitch change pin position for feather and reverse provision; pitch change forces to increase pitch
- 2 code letter for blade design and installation
  - blank = right-hand tractor
  - RD = right-hand pusher
  - L = left-hand tractor
  - LD = left-hand pusher
- 3 propeller diameter in cm
- 4 No. of blade type (contains design configuration and aerodynamic data) according to the certified hub/blade – combinations
- 5 code letter for design changes  
small letter for changes which do not affect interchangeability of blade set  
capital letter for changes which restrict or exclude interchangeability of blade set





**SECTION: ADMINISTRATIVE**

**I. Acronyms and Abbreviations**

n/a

**II. Type Certificate Holder Record**

Avia Propeller Ltd.

**III. Change Record**

<b>Issue</b>	<b>Date</b>	<b>Changes</b>	<b>TC issue</b>
Issue 01	08 February 2018	Initial Issue	Initial Issue, 08 February 2019

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